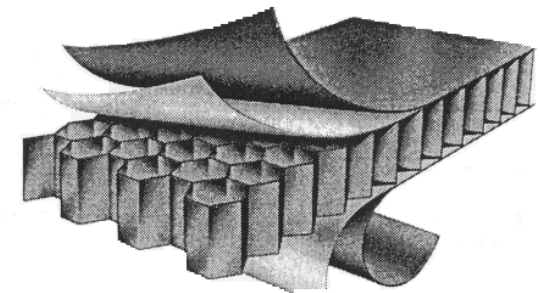
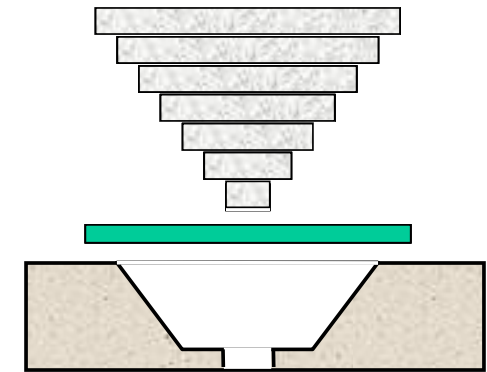
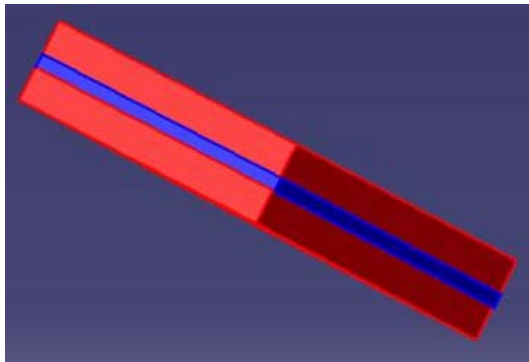


# Improving In-Service Inspection of Composite Structures

## *CACRC Damage Assessment & Quality Inspection Task Group Update*



Dennis Roach  
Sandia National Labs  
FAA Airworthiness Assurance Center



## ITG Team Participants

### CACRC Inspection Task Group Members:

**John Hewitt – Airbus (Co-chair)**

**Jim Hofer - Boeing**

**Jeff Kollgaard – Boeing**

**Kirk Rackow - Sandia Labs AANC**

**Dennis Roach - Sandia Labs AANC (Co-chair)**

**Glae McDonald - US Airways**

**Darrell Thornton – UPS**

**Richard Watkins - Delta Air Lines**

**Bob Stevens – United Airlines**

**Eric Bartoletti – American Airlines**

**Alex Melton - Northwest Airlines**

**Ana Tocalino - Embraer**

***Dave Galella, Al Broz, Rusty Jones, Larry Ilcewicz – FAA***





## CACRC Inspection Task Group Activities

- **Composite NDI Handbook**
  - Complete (SAE ARP5089); requires update
- **Industry wide NDI Reference Standards**
  - Complete (SAE ARP5506 & 5507; DOE report distributed in June 2004)
- **NDI Assessment: Honeycomb Structures**
  - Experiments completed in 2007
  - DOT report in progress
- **NDI Assessment: Solid Laminate Structures**
  - Experiment development completed
- **Miscellaneous Ongoing and Planned Studies**
  - Detection and quantification of weak bonds
  - Affect of porosity, repairs & other impediments on NDI
  - As required to support main tasks & other task groups



# Composite Inspection Reference Standards



- Industry-wide composite reference standards developed to support damage assessment & inspection – honeycomb & solid laminate
- SAE Aerospace Recommended Practices (ARP 5605 & 5606) – adopted into Boeing and Airbus NDT Manuals
- Improve inspections of composite structures via introduction of advanced NDI methods
- Provides consistent approach to composite inspections – harmonized approach by OEMs worldwide



**Optimized NDT Reference Standards**

**Assessed all construction scenarios and determined the variables that affect NDI – final NDI Reference Standards designed accordingly**

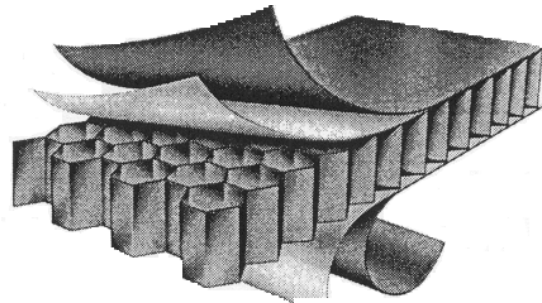




## Composite Honeycomb Flaw Detection Experiment

Utilize airline inspectors to establish industry-wide performance curves that quantify:

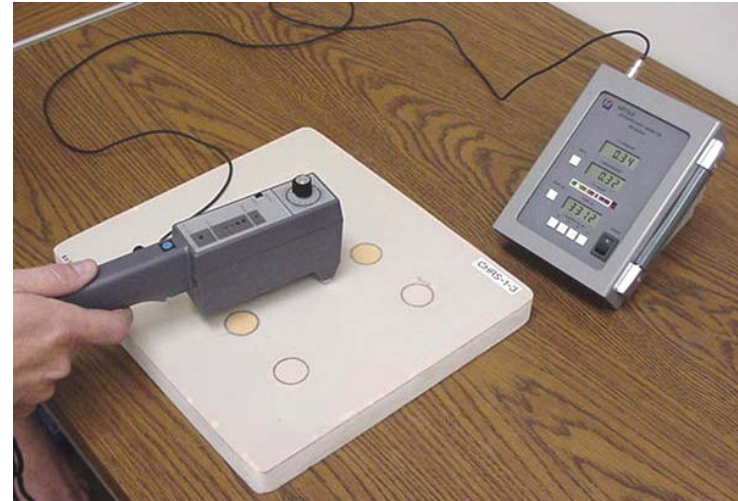
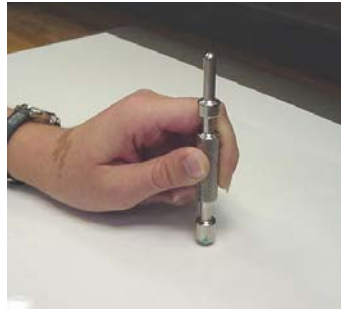
- 1) how well current inspection techniques are able to reliably find flaws in composite structures
  - 2) the degree of improvements possible through the integration of more advanced NDI techniques and procedures.
- Statistically relevant and realistic flaw profiles
  - Blind application of techniques to study hits, misses, false calls, and flaw sizing



# Conventional NDI Devices



**Manual Tap Hammer**



**Automated Tap Hammer**

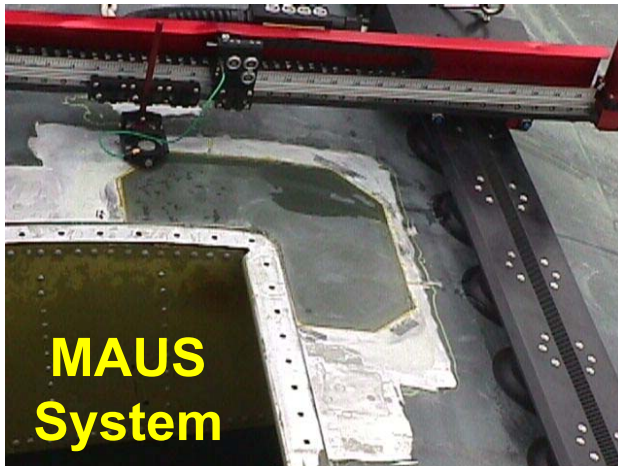
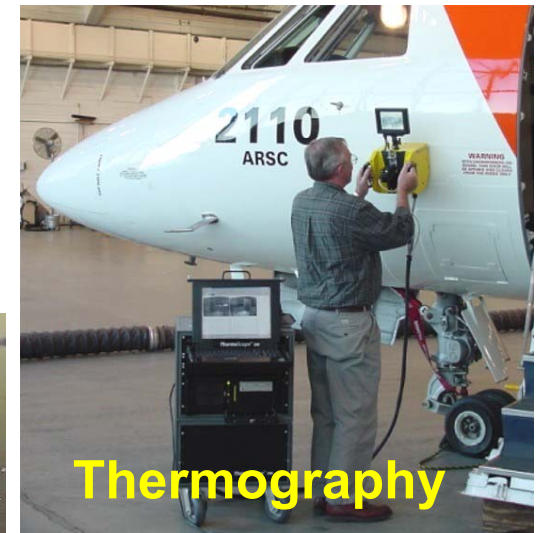


**V-95 Mechanical Impedance Analysis**



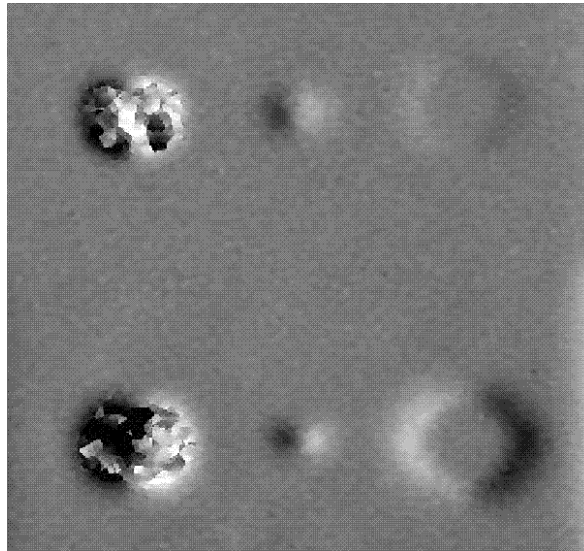
**S-9 Sondicator (LFBT)**

# Wide Area and C-Scan Inspection Methods

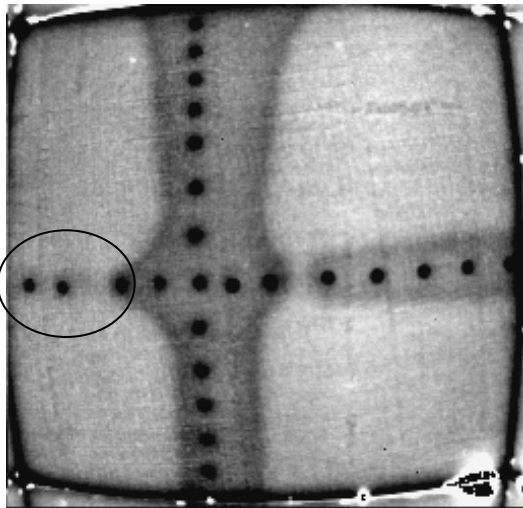




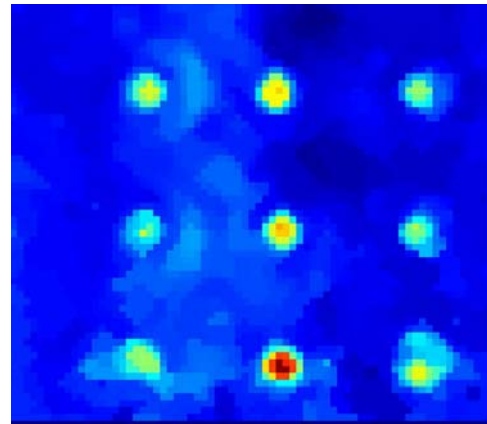
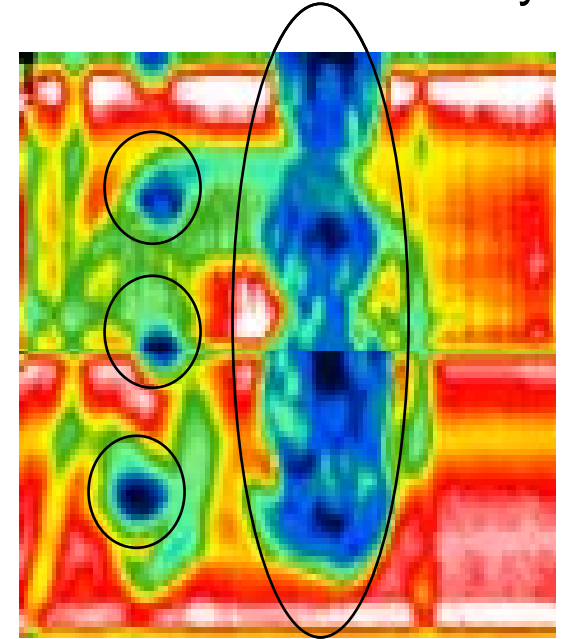
Shearography (LTI) Image



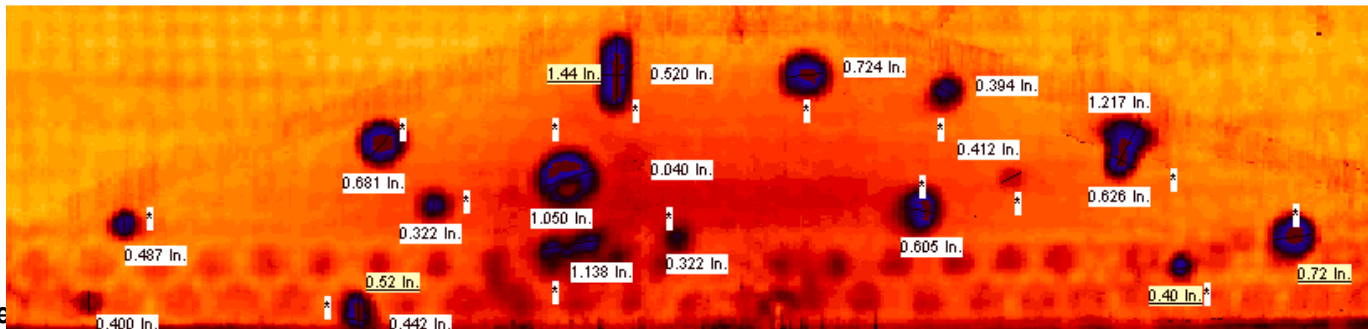
Thermography (TWI) Image



Ultrasonic Wheel Array



SAM Image



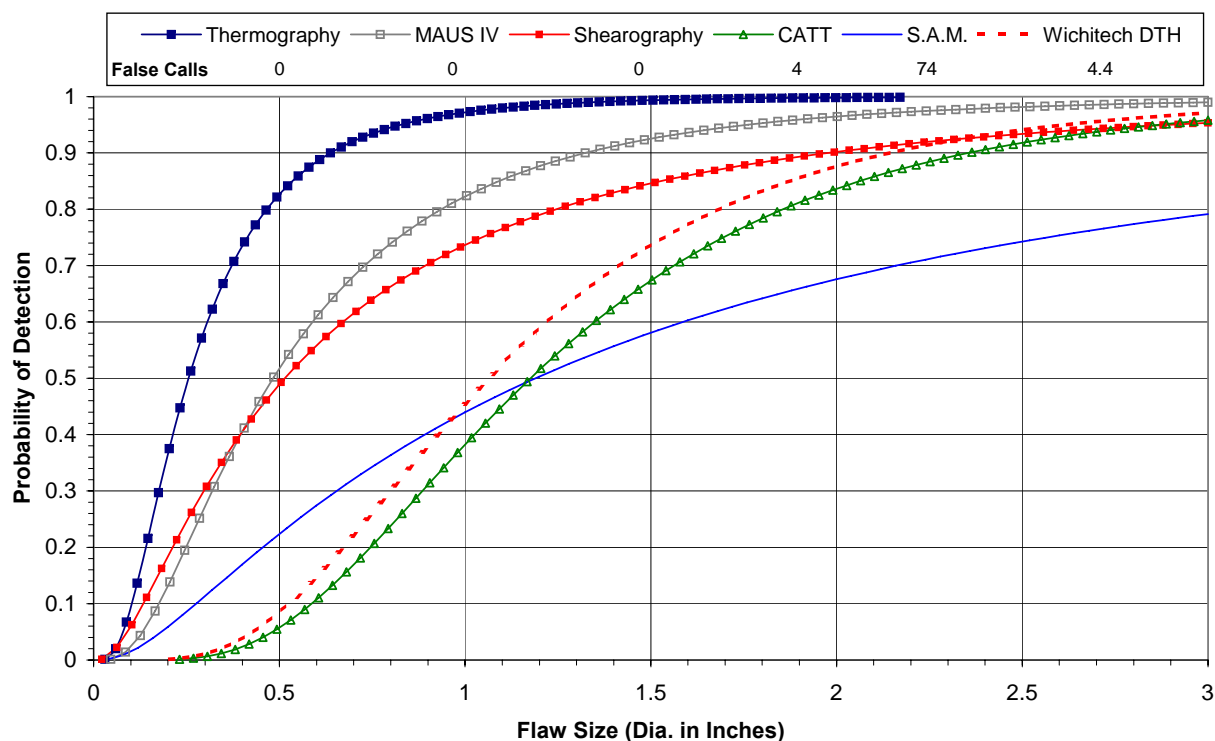
MAUS Image





# Performance of Multiple Devices for A Single Type of Test Specimen

Comparison of Advanced Inspection Techniques with Best Conventional NDI Result on 9 Ply Carbon



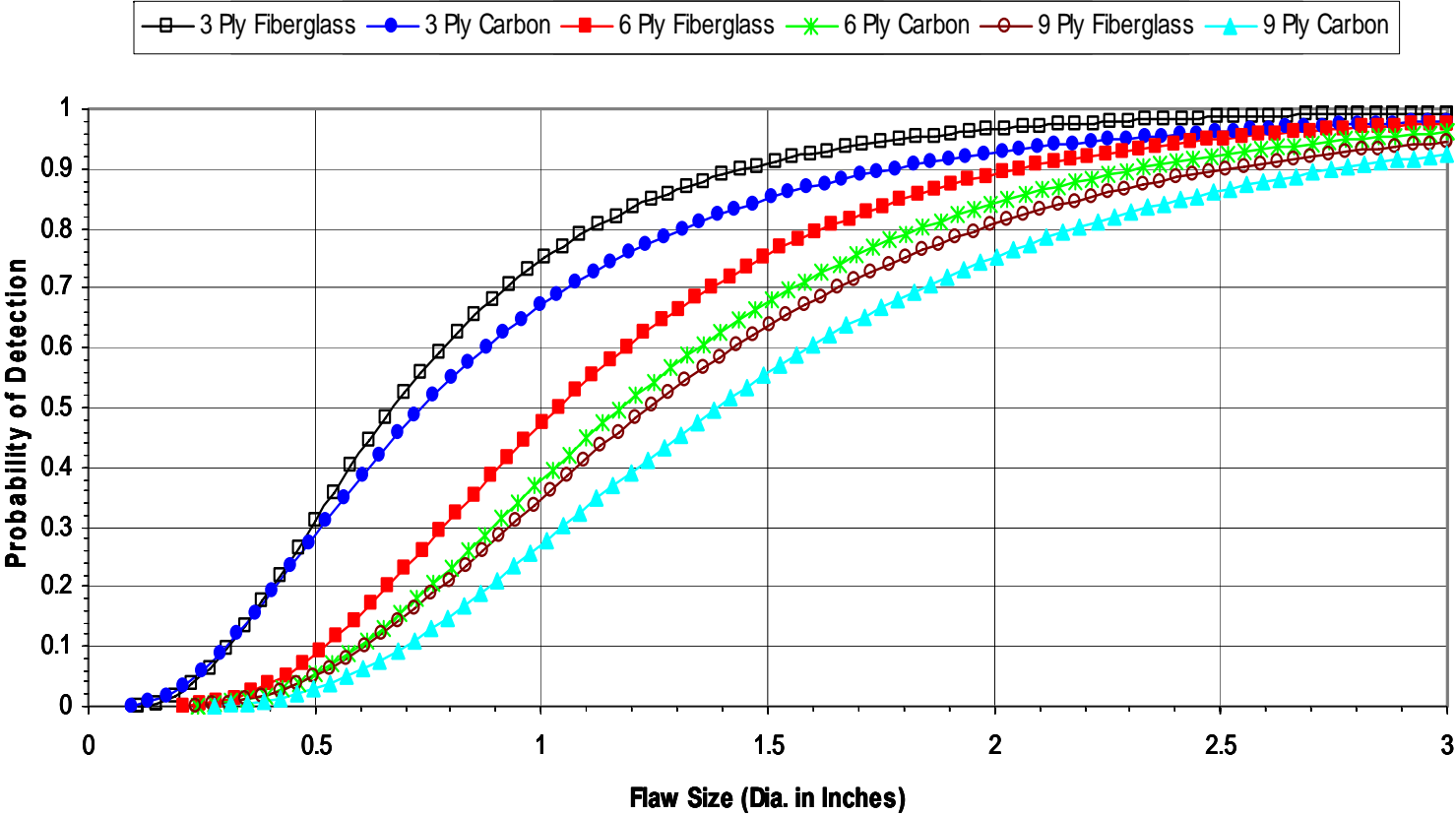
## Results - evaluate performance attributes

- 1) accuracy & sensitivity (hits, misses, false calls, sizing)
- 2) versatility, portability, complexity, inspection time (human factors)
- 3) produce guideline documents to improve inspections
- 4) introduce advanced NDI where warranted



# Performance of Single Device (Woodpecker) Over Range of Test Specimen Types

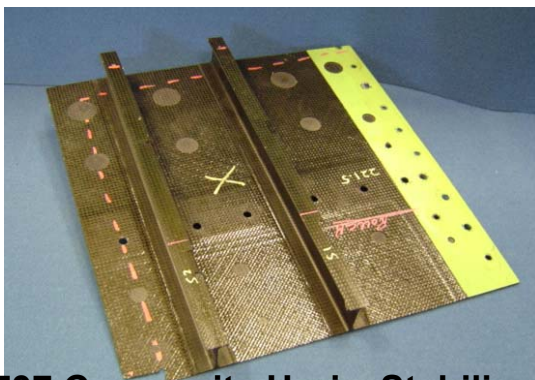
Cumulative PoD - Woodpecker for All Panel Types



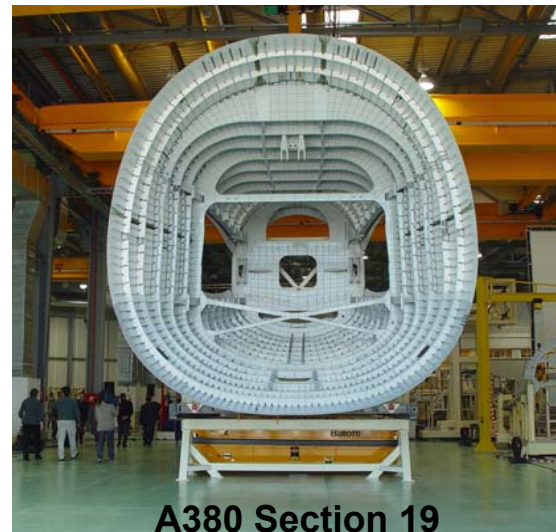
# An Experiment to Assess Flaw Detection Performance in Composite Laminate Structures

## Purpose

- Determine in-service flaw detection capabilities: 1) conventional NDT methods vs. 2) improvements through use of advanced NDT.
- Optimize laminate inspection procedures.
- Compare results from hand-held devices with results from scanning systems (focus on A-scan vs. C-scan and human factors issues in large area coverage).
- Provide additional information on laminate inspections for the “Composite Repair NDT/NDI Handbook” (ARP 5089).

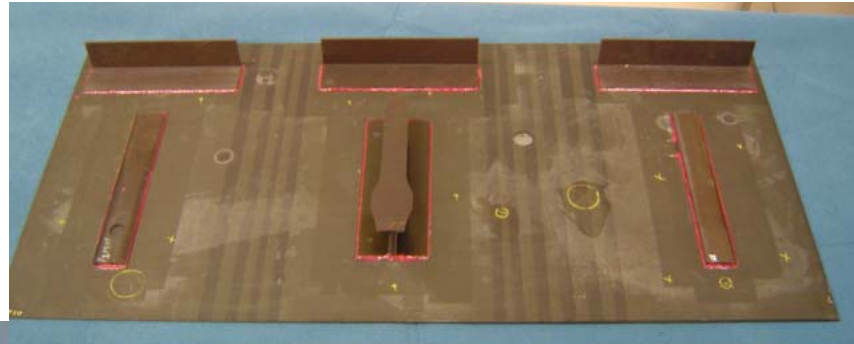


737 Composite Horiz. Stabilizer



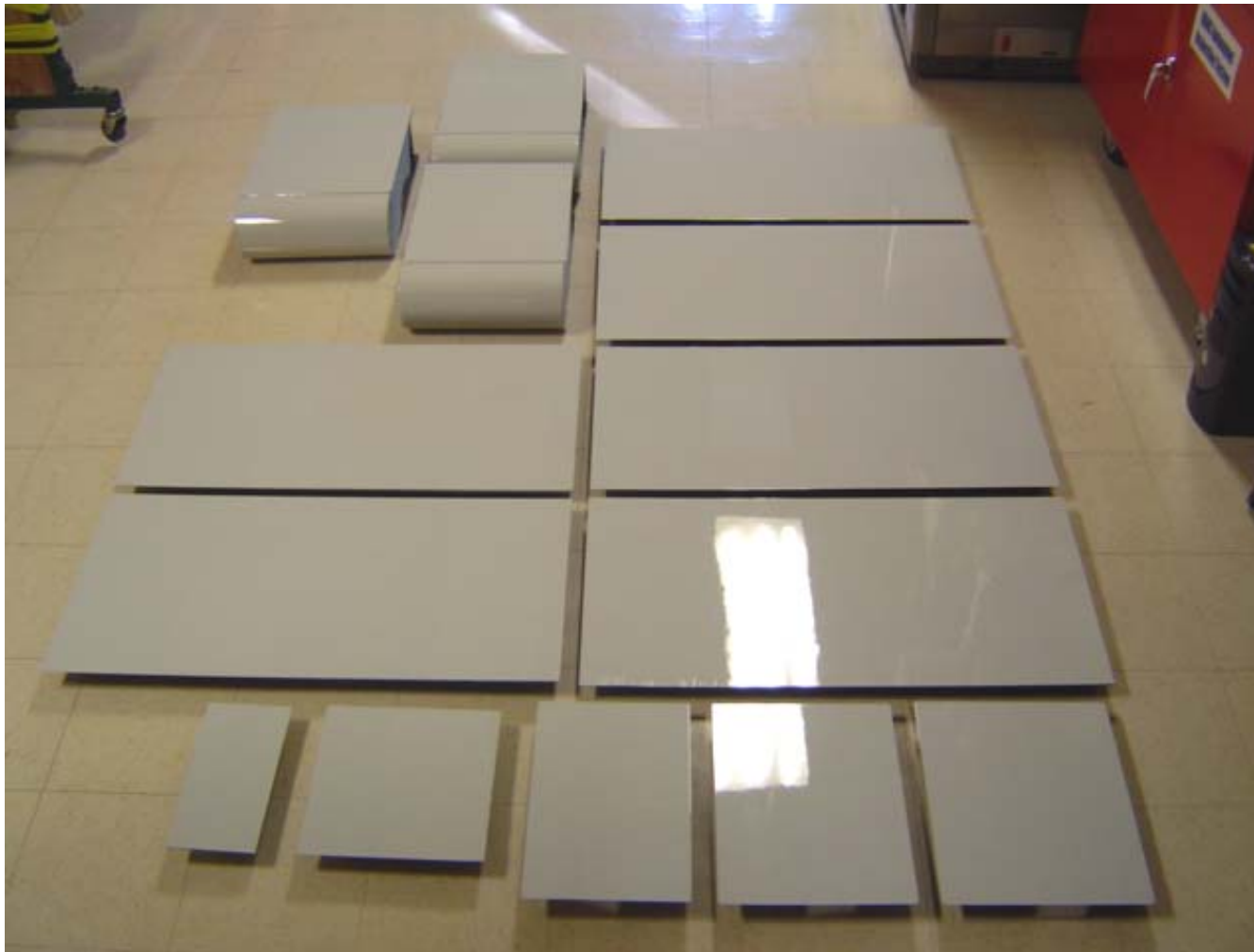
A380 Section 19

# Thick Laminate With Complex & Simple Taper





## Specimen Set - Flaw Detection in Solid Laminate Composites



**Thickness Range:  
12 – 64 plies**

**Simple Tapers**

**Complex tapers**

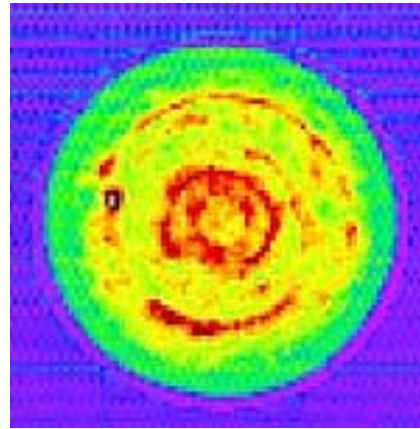
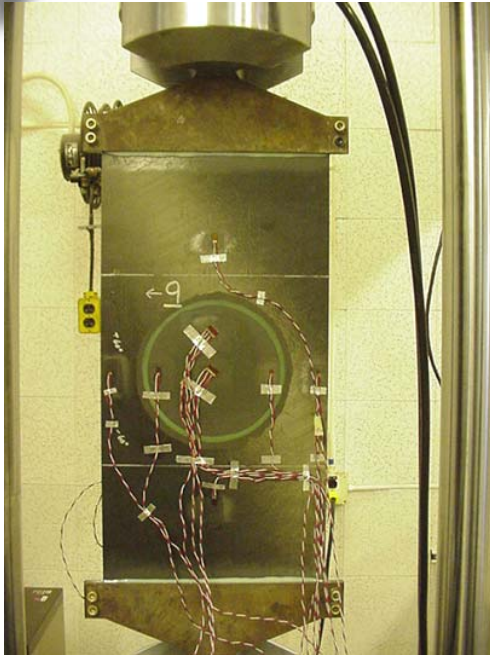
**Substructure Flaws**

**Curved Surfaces**

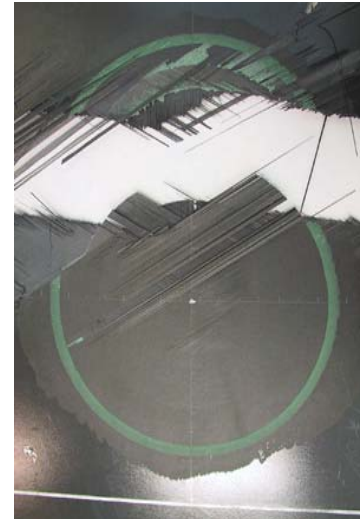
**Array of flaw types**

**NDI Ref. Stds.**

# Composite NDI & Laminate Repair Systems – Compare Mechanical & NDI Performance



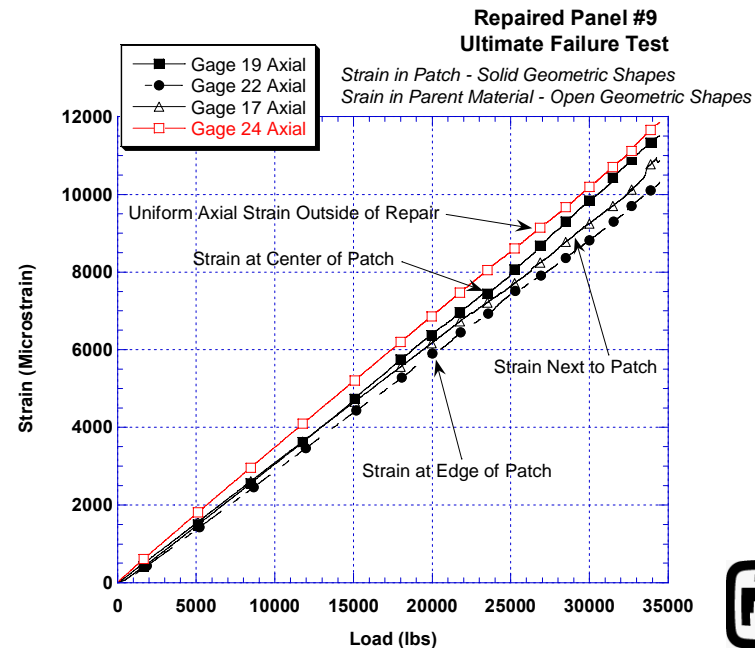
Ultrasonic C-Scan



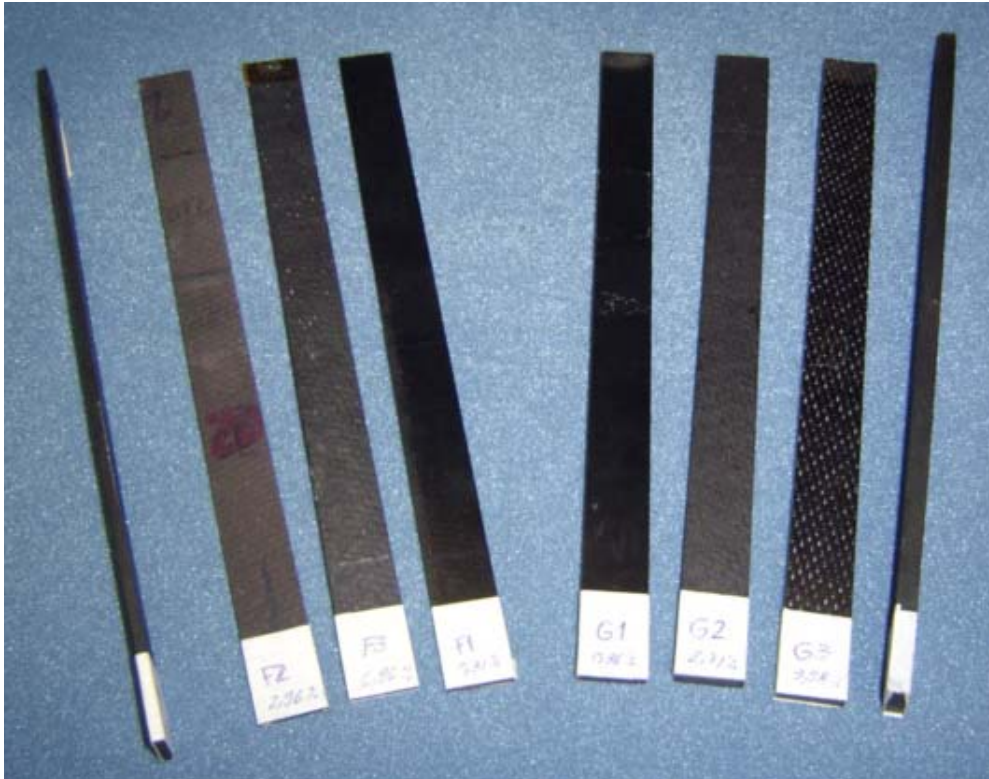
Strain field &  
repair efficiency  
assessment vs.  
NDI findings

Uniaxial carbon  
graphite; plies  
plus one repair

**Comprehensive evaluation of  
composite repair and associated  
NDI technology to ensure proper  
mesh between structural  
integrity & flaw detection**



# Use of NDI to Quantify Porosity Levels & Assess Mechanical Properties vs. Porosity



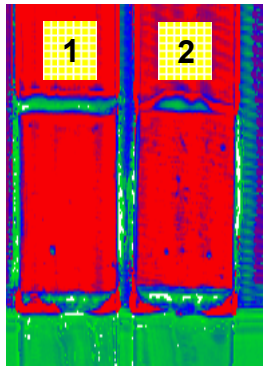
**Carbon Weave**

- Intercompare mechanical tests, NDI & acid etch methods to assess porosity, strength and fatigue life
- Use of NDI tests to calculate mechanical properties
- Use of advanced NDI to improve quantification of porosity (stratified porosity that may exist in a repair)

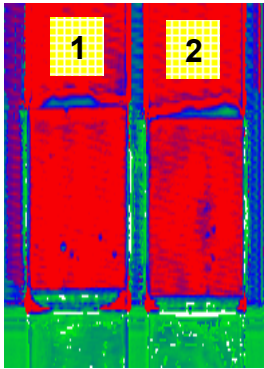


# Characterize Bonded Joints: Quantify Adhesive Strength

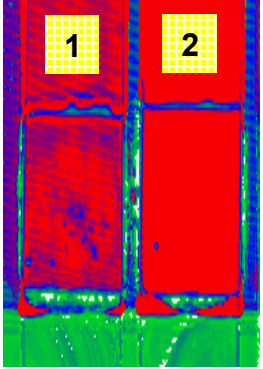
TTU of Weak Bond Specimens Show Trends



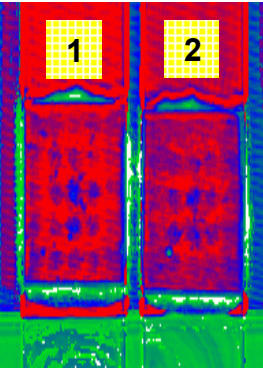
Pristine -  
Best



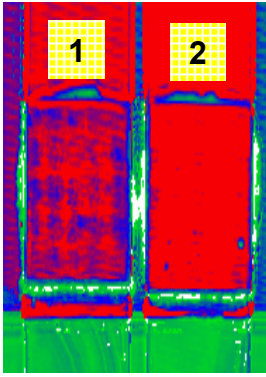
70% MR  
Dilution



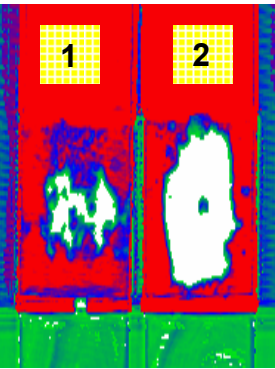
30% MR  
Dilution



33% MR  
Screen



66% MR  
Screen



Room  
Temp.



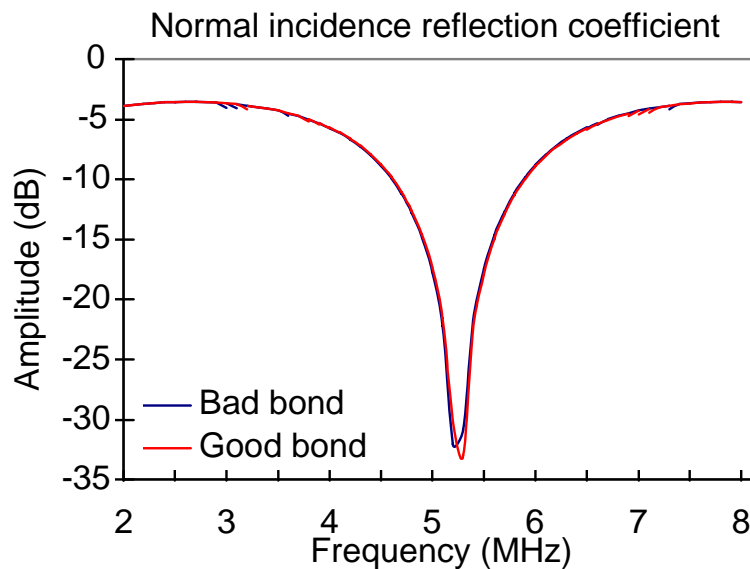
100 MR



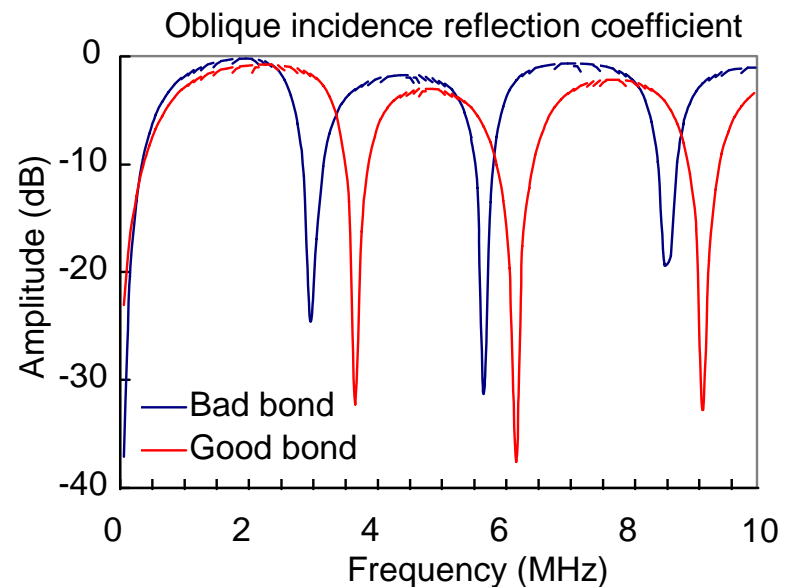


# Angle Beam Ultrasonic Spectroscopy (ABUS)

- Compare received and transmitted waveforms in frequency domain; study frequency/amplitude shifts & change in damping in FRF
- Oblique wave (broadband UT beam) introduces shear stress in the bond line
- Difference between longitudinal wave and shear wave interrogation



**Negligible Affects on  
Normal Wave (?)**



**Frequency and Amplitude  
Shifts Differentiate Bonds**



## Future Direction of *CACRC Damage Assessment & Quality Inspection Task Group*

- **Detection of Barely Visible Impact Damage (BVID); determining extent of composite damage → driven by visual detection**
- **Enhanced visual inspection**
- **Inspection of scarfed composite repairs; in-process QA**
- **Inspection of fastened repairs to composite structure (in-process QA; no current post-repair NDI requirement)**
- **NDI vs. Damage Tolerance vs. Residual Strength – assess structural integrity (focus on particular materials?)**
- **NDI of adhesive bonds (“kissing” disbonds; weak bonds)**
- **Ascertaining deterioration of material properties due to environmental exposure (temp., moisture, chemicals, stress)**





## Future Direction of *CACRC Damage Assessment & Quality Inspection Task Group*

- **Detection of fluid ingress**
- **Quantification of porosity (ref. stds. are critical)**
- **Detection of matrix micro cracking hidden beneath painted surfaces**
- **Rapid, large area inspection methods (improve POD; decrease false calls)**
- **Methods to inspect highly attenuative materials (weaves, thick structures, multi-layered structures) – proprietary issues?**
- **Utilization of SHM techniques**
- **Evaluating NDI performance - assessing conventional NDI in light of advanced NDI methods**



## Future Direction of *CACRC Damage Assessment & Quality Inspection Task Group*

- Optimization of NDI procedures; improved documentation and guidance
- Training – knowledge of hardware & procedures; use of “qualification standards”; industry standardization
- Ramp NDI – qualified personnel; equipment availability
- Implement a database – trends assessment



# *CACRC Inspection Task Group Update and Overview*



Dennis Roach  
Sandia National Labs  
FAA Airworthiness Assurance Center  
(505)844-6078  
[dproach@sandia.gov](mailto:dproach@sandia.gov)

